

$$K_R = \left( E \cdot \sigma_{ys} \cdot \delta \right)^{1/2} \quad (4)$$

where  $E$  and  $\sigma_{ys}$  are the modulus of elasticity and the yield strength of the material being tested, respectively.

## Results and Discussion

### 1. R-Curve Results

1A. Summary of Basic Behavior. The results obtained from all 24 R-curve tests are presented in summary form in Tables IV and V. Table IV gives a complete characterization of each test strictly in terms of LEFM parameters, regardless of whether LEFM or COS analysis was necessary to quantitatively characterize the point of fracture instability,  $K_C$ . Table V gives a summary of all R-curve and  $K_C$  results in terms of the appropriate LEFM or COS method of analysis. The specific method of analysis (LEFM or COS) used for calculating  $K_C$  is listed for each specimen in Table V. The basis for the choice of analysis at  $K_C$  is given in Table IV. The results from all 24 R-curve tests will be discussed in separate sections below.

### 1B. Effects of Temperature for B = 1.5-Inch Specimens.

The individual R-curves obtained for all B = 1.5-inch (38 mm) specimens of the 50 ksi steel tested at nominal temperatures of -40, +40, and +72 F are presented in Figures 4, 5, and 6, respectively. The two R-curves obtained for the 62-ksi steel are presented in Figure 7. The crack extension,  $\Delta a$ , shown in all such R-curve plots

of the present study corresponds to actual crack extension ( $\Delta a = \Delta a_{act}$ ) as determined with double-compliance calibration procedures.

The results in Figures 4 through 6 show, collectively, that a rapid increase in plane-stress crack tolerance occurs with increasing temperature. Evidence for this can be seen in both the increasing amounts of stable crack extension preceding fracture,  $\Delta a_c$ , and the increasing  $K_c$  values that occur with increasing temperature. The specific variation in  $K_c$  values with temperature for all the B = 1.5-inch specimens tested is presented in a summary plot, Figure 8. This figure shows that the  $K_c$  transition is quite steep at temperatures above 0 F (-18 C).

Figure 8 also shows that, within the limitations of evaluation based on only two specimens, the  $K_c$  behavior of the 62-ksi steel is the same as that for the 50-ksi steel. This result is surprising to some extent, since the static  $K_{IC}$  transition temperatures for these same two steels are somewhat different.<sup>10)</sup>

Comparisons of  $K_c$  and  $K_{IC}$  for each of the two steels will be treated in detail in a separate section (2F) below. It is sufficient for the present to note that a "pop-in" behavior was observed for the 62-ksi steel at +72 F. This classical R-curve behavior for the 62-ksi steel was unique in the present study and occurred at a  $K_R$  value of approximately 120 ksi  $\sqrt{\text{inch}}$  (132 MNm<sup>-3/2</sup>) compared with complete fracture at a  $K_c$  value of 365 ksi  $\sqrt{\text{inch}}$  (400 MNm<sup>-3/2</sup>), a level 3 times higher.

Several additional features should be noted in the summary plot of  $K_{Ic}$  behavior, Figure 8. The first is the existence of a relatively wide "scatter band" in the  $K_{Ic}$  results obtained. This is an important consideration that will be discussed separately in a subsequent section (2A). The second feature is the extremes in behavior exhibited by the 2T and 7C specimens of the 50-ksi steel tested at +72 F. That is, the 2T specimen fractured at an apparent  $K_{Ic}$  value in excess of  $87 \text{ ksi } \sqrt{\text{inch}}$  ( $96 \text{ MNm}^{-3/2}$ ), whereas the 7C specimen did not fail at a  $K_{Ic}$  value of  $477 \text{ ksi } \sqrt{\text{inch}}$  ( $525 \text{ MNm}^{-3/2}$ ), the limit of the test-machine capacity, Table V.

The result for the 7C specimen tested at +72 F is indicative of true material behavior, but the result for the 2T specimen tested at +72 F is not. Rather, the 2T specimen result is a spurious reflection of the method of testing and analysis. Because of the critical nature of this point these results must be described in some detail. First of all, the 2T specimen result at +72 F can be seen to be clearly inconsistent with the results obtained from the same 2T size specimens tested at lower temperatures (-40 and +40 F), Figure 8. The untypical nature of the result at +72 F resulted primarily from limitations in the ability to accurately analyze  $K_{Ic}$  at fracture. That is, both of the 2T specimens tested at lower temperatures exhibited relatively large amounts of plasticity and required analysis by COS, Tables IV and V. On the basis of these lower temperature results, extensive plasticity and deviation from linearity in the  $P$  vs  $V_1$

test record were expected at +72 F. However, the 2T specimen at +72 F fractured prematurely at a 15.5 percent secant intercept value in the P vs  $V_1$  test record prior to the first scheduled partial unloading step--thereby precluding meaningful COS analysis at fracture instability. Despite the nearly linear nature of the P vs  $V_1$  test record, the  $K_I$  value at fracture calculated on the basis of LEFM (the only alternative analysis method) was also not meaningful because it was substantially in excess of the various universally accepted limits for LEFM calculations. In particular, as shown in Table IV, the value at fracture for the 2T specimen at +72 F calculated on the basis of LEFM,  $K_{I,max} = 91.3 \text{ ksi}\sqrt{\text{inch}}$ , was well above both the limit for plane-strain calculations,  $K_{I,Lub} = 38.9 \text{ ksi}\sqrt{\text{inch}}$ , and each of the conservative limits for plane-stress calculations  $K_{MC} = 53.3 \text{ ksi}\sqrt{\text{inch}}$  and  $K_{BSY} = 69.7 \text{ ksi}\sqrt{\text{inch}}$ . When conditions are such that these LEFM limits are substantially exceeded due to material behavior--as in the present case--calculations based on LEFM have no physical significance because they grossly underestimate the true material behavior in terms of  $K_I$ .<sup>10,13,14</sup> This is, of course, the exact reason why COS analysis of such elastic-plastic fracture behavior is necessary for tough materials. However, the value at fracture for the 2T specimen at +72 F calculated on the basis of COS using specific approximations,  $K_R \geq 87 \text{ ksi}\sqrt{\text{inch}}$ , Table V, represents a lower bound that is even less than that based on LEFM ( $K_{I,max} = 91.3 \text{ ksi}\sqrt{\text{inch}}$ ) and thus even less meaningful. The reasons for this behavior are currently unknown.

Thus, the conclusion is that the  $K_c$  value for the 2T specimen tested at +72 F is invalid, and accordingly, that the results (calculated on the basis of either LEFM or COS) are suppressed in a manner similar to that in which invalid  $K_{IC}$  ( $K_Q$ ) results become suppressed when specimen dimensions are inadequate.<sup>10,13,14</sup> This artificially lower  $K_c$  result from COS appears to be a specific consequence of the violation of both the minimum specimen size and the minimum specimen proportions required for valid COS results (presently undefined).

The greater resistance to fracture exhibited by the A572 steels with increasing temperature in Figure 8 is also confirmed by examining the fracture surfaces of the 2T and 4T CT specimens (tested under load-control conditions), Figures 9 and 10, respectively. The size of the 2T and 4T specimens tested at +72 F in relation to the size of the corresponding 7C specimen tested at the same temperature is illustrated in Figure 11. This figure also illustrates the extensive crack tolerance exhibited by the 7C specimen at the point where the test was terminated without specimen failure ( $\Delta a \approx 0.60$  inch or 15 mm on the specimen surface at  $K_R = 477 \text{ ksi } \sqrt{\text{inch}}$  or  $525 \text{ MNm}^{-3/2}$ ). The  $K_c$  value for this 7C specimen of the 50-ksi steel tested at +72 F is thus in excess of  $477 \text{ ksi } \sqrt{\text{inch}}$ . This compares with a  $K_c$  value of  $365 \text{ ksi } \sqrt{\text{inch}}$  ( $400 \text{ MNm}^{-3/2}$ ) for the 7C specimen of the 62-ksi steel that was tested at +72 F.

Additional support for the conclusion that the  $K_c$  result for the 2T specimen is invalid can be seen from the fracture surfaces of

the specimens tested at +72 F, Figure 11. Specifically, the nature of the stable crack extension for the 7C specimen was such that a "shear lip" (45-degree or 0.77 rad slant fracture) began to develop directly from the tip of the original fatigue crack at the specimen surface. A similar shear lip also began to develop for the 4T specimen but only after the original fatigue crack had extended on the expected crack plane by a small amount (1/8 to 1/4 inch, or 3.2 to 6.4 mm). However, the fracture surface for the 2T specimen was completely flat, without any evidence of stable shear-lip formation on the specimen surface prior to fracture. The fracture surface for the 2T specimen was the same as that obtained earlier for similar-size specimens tested in 3-point bending at the same +72 F temperature in attempts to measure static  $K_{IC}$ .<sup>10)</sup> In both cases, the fracture surfaces were completely flat because of the constraining influence of the limited specimen ligament ( $W - a = 2.30$  to 2.70 inches, or 58 to 69 mm), or specifically, the close proximity of the specimen back surface to the crack tip. Similarly, in both cases, the calculated  $K_{IC}$  or  $K_C$  values (both calculated on the basis of LEFM) were invalid and exhibited substantial  $K_I$ -suppression effects.<sup>10,13,14)</sup> The  $K_{IC}$  behavior and extent of  $K_I$ -suppression effects for the invalid  $K_Q$  results of both steels are discussed more fully below (Section 2F).

1C. Effects of Temperature for B = 0.5-Inch Specimens. The individual R-curves obtained for the B = 0.5-inch (12.7 mm) 2T and 4T specimens of the 50-ksi A572 Grade 50 steel tested at nominal temperatures of -40, +40, and +72 F are presented in Figures 12, 13, and 14, respectively. As with the earlier results for the B = 1.5-inch specimens, these results for the B = 0.5-inch 2T and 4T specimens are consistent in showing increasing amounts of stable crack extension preceding fracture,  $\Delta a_c$ , with increasing test temperature.

However, the corresponding variation in the  $K_c$  values with increasing temperature for the 2T and 4T specimens appears to be quite different, Figure 15. Specifically, the  $K_c$  values obtained with the 2T specimens appear to be insensitive to temperature, since all values were essentially in the range  $300 \pm 20 \text{ ksi } \sqrt{\text{inch}}$  ( $330 \pm 22 \text{ MNm}^{-3/2}$ ). These results are in contrast to the results from the 4T specimens, which indicate a strong sensitivity to temperature, with the  $K_c$  behavior increasing from  $150 \text{ ksi } \sqrt{\text{inch}}$  ( $165 \text{ MNm}^{-3/2}$ ) at -40 F (-40 C) to approximately 400 to 500  $\text{ksi } \sqrt{\text{inch}}$  (440 to 550  $\text{MNm}^{-3/2}$ ) at +72 F (+22 C). These different trends in behavior for the two different size specimens are reflected in the resulting fracture surfaces. In particular, all the 2T specimens tested show approximately the same amount of stable crack extension ( $\Delta a$ ) on the actual fracture surfaces prior to fracture instability at  $K_c$ ,

Figure 16, whereas the fracture surfaces of the 4T specimens show a corresponding marked sensitivity to temperature, Figure 17. The high degree of fracture toughness exhibited by the 4T specimens tested at +72 F can be seen in Figure 18.

1D. Nature of the Fracture-Instability Event. Except for three of the 24 specimens tested in the present study, all fractures occurred at  $K_{IC}$  and in a catastrophic manner. Furthermore, this behavior was observed regardless of whether the tests were conducted under load-control or displacement-control conditions. The suddenness of the complete-fracture event at  $K_{IC}$  may, at first, be somewhat unexpected for the tests conducted at the higher temperatures, particularly for those specimens tested under displacement-control conditions (maximum crack stability). However, such results are more easily understood when it is considered that this behavior is merely a reflection of the inherent strain-rate sensitivity (to fracture) for this steel, and therefore, for all steels of this same strength level (50 ksi or  $345 \text{ MN/m}^2$ ), since strain-rate sensitivity of fracture behavior depends primarily on  $\sigma_{ys}$ .<sup>10)</sup> Furthermore, this behavior is again less surprising when it is considered that even the toughest steels have limited ductility, and therefore fail in a similar sudden manner when tested in a conventional tension test.

The three exceptions to the behavior described above were all tested at +72 F (+22 C): the B = 1.5-inch 7C specimen and the

duplicate B = 0.5-inch 4T specimens. This exceptional behavior occurred with the largest specimens tested at each thickness. The behavior obtained from these specimens with the larger in-plane dimensions is a more accurate measure of the intrinsic plane-stress crack tolerance for each thickness of A572 Grade 50 steel than the behavior obtained from the corresponding smaller specimens. Stated differently, the larger in-plane dimensions for these specimens allow measurements of the true R-curve to higher  $K_R$  levels before the results became biased because of violation of the presently undefined limits of COS validity. This reasoning leads to the obvious conclusion that the largest specimen size compatible with testing capabilities should be used in R-curve evaluations of high-toughness materials when it is evident that COS procedures are necessary.

As described previously, the reason that complete fracture did not occur for the B = 1.5-inch 7C specimen tested at +72 F was that the deflection limits of the displacement-control testing machine were exceeded. Thus, the  $K_c$  value for this 7C specimen was in excess of the  $K_R = 477 \text{ ksi } \sqrt{\text{inch}}$  ( $525 \text{ MNm}^{-3/2}$ ) value attained at test termination and, as discussed earlier, the inconsistent result obtained from the corresponding 2T specimen (apparent  $K_c = 87 \text{ ksi } \sqrt{\text{inch}}$  or  $95 \text{ MNm}^{-3/2}$ ) can be dismissed as an invalid result occurring because of violation of COS requirements.

A similar type of influence can be seen in analysis of the B = 0.5-inch specimens tested at +72 F. These include the duplicate 4T specimens that exhibited exceptional behavior in not fracturing catastrophically at the cited  $K_C$  values. Specifically, these two 4T specimens yielded  $K_C$  values of >380 and >503 ksi  $\sqrt{\text{inch}}$  (>420 and >550  $\text{MNm}^{-3/2}$ ), values of  $K_R$  that occurred at increments of stable crack extension,  $\Delta a$ , of 1.10 and 0.90 inches (28 and 23 mm), respectively, Figure 14. The extreme crack tolerance exhibited by these 4T specimens can be seen by the fact that complete catastrophic fracture occurred for each of these specimens only after significantly greater increments of stable crack extension, specifically, values of  $\Delta a_C = 3.95$  and 3.47 inches (100 and 88 mm), respectively, Figure 17, and then only under the action of significantly higher crack-tip strain rates (intentional fracture). However, the above values were cited for  $K_C$  because subsequent calculations made for  $\Delta a$  values beyond 1.10 and 0.90 inches led to lower values of  $K_R$ , an unrealistic assessment of true plane-stress fracture behavior. Thus, although the true  $K_C$  values for the 4T specimens are clearly greater than the cited values of  $K_R$ , subsequent results for each of these specimens (because they are lower) again represent a clear violation of the presently undefined requirements for valid COS results. Such violations are not unexpected when it is considered that they occurred well beyond the attainment of the maximum load point ( $P_{\text{max}}$ ), a value that represents limit-load or full-plastic-hinge conditions, Figure 19.

The above clear violation of COS requirements evident in the  $K_{IC}$  results for the duplicate 4T specimens tested at +72 F indicates that there may be a similar influence in the specific  $K_{IC}$  result obtained for the corresponding 2T specimen tested at +72 F ( $K_{IC} = 308 \text{ ksi } \sqrt{\text{inch}}$  or  $340 \text{ MNm}^{-3/2}$ ). The striking difference in the extent of the stable crack extension preceding fracture ( $\Delta a_c$ ) evident on the fracture surfaces of these 2T and 4T specimens, Figures 16 and 17, would support this contention. Furthermore, the insensitive nature of the observed  $K_{IC}$  values ( $300 \pm 20 \text{ ksi } \sqrt{\text{inch}}$ , or  $330 \pm 22 \text{ MNm}^{-3/2}$ ) over the entire temperature range from -40 to +72 F (-40 to +22 C), described earlier, is in sharp contrast to that expected on an intuitive basis and thus provides an additional indication of such a possibility. However, insufficient test results are available to indicate the extent to which the 2T specimen result may be influenced as a result of violation of COS requirements.

## 2. General Discussion

The influence of any parameter on the R-curve behavior of a given material can be measured in terms of any of the three principal characteristics of an R-curve: (1) the  $K_{R}$  value at the onset of stable crack growth, (2) the increment of stable crack extension at fracture instability,  $\Delta a_c$ , or (3)  $K_{IC}$ . If the stable-crack-growth characteristics are neglected, the influence of any parameter on R-curve behavior can be reduced to a direct comparison of the resulting  $K_{IC}$  values. It is on this basis that assessments of the influence of various parameters on R-curve behavior are described below.

2A. Overall Scatter Observed in  $K_C$  Results. When plane-stress fracture tests are conducted under fixed material and test conditions ( $T$ ,  $\epsilon$ , and  $B$ ) by using specimens with different initial crack lengths,  $a_o$ , some "scatter" in the resulting  $K_C$  behavior will occur. Specifically,  $K_C$  will increase with increasing crack length,  $a_o$ , for a well-behaved, homogeneous material (" $K_C$  ordering"). Such behavior is, of course, the fundamental basis of R-curve characterization that was described in an earlier section of the present paper (see Figure 1). Such systematic variations in  $K_C$  with  $a_o$  are not real scatter at all, but rather the typical plane-stress fracture behavior that would normally be expected for any material. It is only the deviations from this systematic pattern of  $K_C$ -ordering behavior that can truly be referred to as scatter.

In the present work, the 2T, 4T, 4C, and 7C specimens tested had initial crack lengths,  $a_o$ , of nominally 1.75, 3.00, 3.45, and 5.80 inches (44, 76, 88, and 147 mm), respectively. Accordingly, the  $K_C$  value for a 7C specimen would be expected to be much higher than that for a 2T specimen ( $K_C$  ordering). Similarly, the  $K_C$  values for the 4T and 4C specimens would be expected to be intermediate between these extremes, with very little difference expected between the  $K_C$  values for the 4T and 4C specimens since the  $a_o$  values differ by only a small amount.

Deviations from this normal pattern of plane-stress fracture behavior (true scatter) can occur for a number of reasons. When more

than one test method is used, as in the present work, the extent of such deviations from the normal  $K_C$  vs  $a_0$  behavior depends on (1) the repeatability of results from a single test method, (2) differences between test methods, and (3) variations in fracture toughness of the material tested.

The influence of each of these sources of true  $K_C$  scatter is described in subsequent sections relative to the summary of  $K_C$  results obtained for the  $B = 1.5$ -inch and  $B = 0.5$ -inch specimens, as given in Figures 8 and 15, respectively. However, prior to such analysis several comments are necessary. First, item 1 above involves both the repeatability of the testing conditions and the material variation. That is, items 1 and 3 are related and cannot be isolated entirely from each other. Second, the list given above does not include apparent  $K_C$  results obtained under conditions for which the basic stress analysis is violated, such as the violation of COS requirements described in a previous section, since such results represent an artifact and not true  $K_C$  scatter.

2B. Repeatability of  $K_C$  Results for a Specific Test Method.

The repeatability of  $K_C$  results, or the lack thereof, can only be measured by a direct comparison of individual test results obtained under the same conditions of specimen size, test temperature, and test technique (load-control or displacement-control techniques). If only duplicate specimen tests are available for such purposes, as in the present study, conclusions based on such a small number

of results must clearly involve reservations. However, no alternative choice exists for making an assessment of repeatability in the limited results of the present study on A572 Grade 50 steel.

Four different sets of duplicate specimen tests are available for such assessments in the present study. In particular, Table VI shows that the variations in the average  $K_C$  values obtained from duplicate 2T, 4T, 4C, and 7C specimens are  $\pm 7$ ,  $\pm 14$ ,  $\pm 12$ , and  $\pm 29$  percent, respectively. These variations cannot be related to systematic changes in specimen size because the concomitant test conditions (thickness and test temperature) for each specimen type were different, Table VI. The results in Table VI suggest that singular  $K_C$  variations of less than  $\pm 30$  percent relative to the average  $K_C$  value obtained for duplicate specimens of A572 Grade 50 steel would be expected for tests conducted using either the load-control or the displacement-control testing techniques.

By comparison, the variation in  $K_{IC}$  values for specially melted high-strength steels exhibiting good homogeneity, such as 18Ni(250 Grade) maraging steels, has been shown earlier to be within  $\pm 5$  or  $\pm 10$  percent, depending on the total number of specimens used and the participating laboratories.<sup>15,16)</sup> Furthermore, the variation in  $K_{IC}$  values for lower strength steels exhibiting both less homogeneity and a  $K_{IC}$  transition behavior has been shown to be as large as  $\pm 25$  percent or more for a given temperature and strain rate.<sup>10,14,17,18,19,20)</sup> Thus, the presently observed variation of

as much as  $\pm 29$  percent in one case for the  $K_c$  behavior of A572 Grade 50 steel, a similar low-strength steel, is not surprising. That is, despite the different fracture modes ( $K_c$  vs  $K_{Ic}$ ), the present variation in results ( $K_c$  repeatability) appears to be no greater than that ( $K_{Ic}$  repeatability) observed earlier in similar low-strength steels.

### 2C. Load-Control vs Displacement-Control Test Methods.

To assess the influence of the testing procedure (load-control vs displacement-control test methods) on the  $K_c$  values obtained, it is necessary to compare specific results obtained with each procedure for specimens of the same size (that is, with the same W and B dimensions). Results from the same size specimen for both test procedures are necessary in order to exclude any additional influence of crack length,  $a_o$ , on the  $K_c$  value.

Such a basis of comparison is available from the results of the 4T specimens tested under load-control conditions and the 4C specimens tested under displacement-control conditions, listed as items No. 4 through 10 in Table V. The minimal difference in the initial crack lengths,  $a_o$ , for the 4T ( $a_o = 3.00$  inches) and 4C specimens ( $a_o = 3.45$  inches) can be discounted as second-order effects in such comparisons. A comparison of the  $K_c$  values for the 4T and 4C specimens is given in Table VII. These limited results show that there is a definite influence of the test procedure. In particular, the 4T specimens tested under load-control

conditions at nominal temperatures of -40, +40, and +72 F exhibited  $K_C$  values that were higher than the corresponding  $K_C$  values obtained under displacement-control conditions by 51, 80, and 40 percent, respectively. Although such direct comparisons are admittedly limited in number, they do nevertheless provide a consistent behavior. That is, these results indicate that the  $K_C$  value obtained with the load-control procedure is, on the average, 57 percent higher (1.57 factor) than that for the same size specimen tested using the displacement-control procedure. Furthermore, the higher  $K_C$  values for the 4T specimens tested under load-control conditions are considerably in excess of the maximum observed variation of  $\pm 29$  percent that might be expected strictly on the basis of repeatability (as discussed in the previous section).

While this influence of testing method on the  $K_C$  result appears to be real, it cannot be fully verified using statistical analysis procedures. That is, the present results on  $K_C$  repeatability (previous section) cannot be used to meaningfully assess the standard deviation ( $\sigma$ ) for the variability in  $K_C$  because both the basic nature of the variability cannot be accurately ascertained\* and the total number of duplicate tests (4 sets) available are insufficient in number. The standard deviation for the variability in  $K_C$  must be

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\* Statistical tests to determine the fundamental character of the present variability in  $K_C$  values are inconclusive. That is, it is not known with confidence if the variation is constant (in absolute terms) and independent of  $K_C$  level, or whether the variation is proportional to the mean  $K_C$  level.

known accurately before confidence levels (67 percent for  $\pm\sigma$  and 95 percent for  $\pm 2\sigma$ ) can be established in relation to the statistical assessment of the influence of testing procedure on  $K_C$ .

However, careful inspection of all the  $K_C$  results for the  $B = 1.5$ -inch specimens in Figure 8 also reinforces the basic conclusion that there is an influence of the test procedure. In particular, the  $K_C$  values obtained for the 2T and 4T specimens tested under load-control conditions generally fall in the upper half of the "scatter" band, whereas the  $K_C$  values for the 4C and 7C specimens tested under displacement-control conditions generally fall in the bottom half of the scatter band.

The specific cause of the consistent differences between the  $K_C$  values obtained with the load-control and displacement-control techniques is unknown. Furthermore, such differences were unexpected and in sharp contrast to the earlier results by Heyer and McCabe<sup>9)</sup> which demonstrated complete equivalence between the two loading techniques for both high-strength aluminum and titanium alloys. Specifically, these earlier results showed, for each of a range of eight (8) different material conditions, that  $K_C$  values determined directly with the load-control technique differed by less than  $\pm 5$  percent from the corresponding  $K_C$  values determined with the displacement-control technique.

These earlier results of Heyer and McCabe were all obtained on nonferrous materials in thin sheet form ( $B = 0.066$  inch or

1.7 mm or less); the fracture properties ( $K_{IC}$ ) of the sheet are quite reproducible and primarily independent of strain rate, so that the results were analyzed under completely LEFM conditions. In contrast, the present results on A572 Grade 50 steel are from thick plate ( $B = 1.5$  inches); the fracture properties ( $K_{IC}$ ) of the plate are of questionable reproducibility (see previous and subsequent sections) and highly dependent on strain rate (as indicated by earlier  $K_{IC}$  behaviors), and the fracture toughness ( $K_{IC}$ ) is so high, that the results must be analyzed under COS conditions. These five (5) primary differences between the earlier studies of Heyer and McCabe and the present studies are summarized in Table VIII. Such differences in testing conditions can be combined with consideration of strain-hardening characteristics to provide a salient starting point for the future research work that is necessary in order to understand the reasons for the differences observed in the present  $K_{IC}$  results on A572 Grade 50 steel as determined by the load-control and displacement-control techniques.

2D. Effects of Thickness ( $B = 1.5$  inch vs  $B = 0.5$  inch) for the Load-Control Test Method.

Because the previous section showed that the  $K_{IC}$  results are influenced by testing method (load control vs displacement control) an assessment of the influence of specimen thickness ( $B = 1.5$  inch vs  $B = 0.5$  inch) can only be made in a meaningful manner using a single test method. Such comparisons are available from the load-control tests conducted using both 2T and 4T specimens for each of the  $B = 1.5$ -inch and  $B = 0.5$ -inch thicknesses. The results from

these direct comparison tests are summarized at each of the three different test temperatures in Table IX.

The specific results for the 2T specimens in Table IX show that there is an effect of thickness, with higher  $K_{IC}$  values consistently being obtained for the thinner ( $B = 0.5$  inch) specimens than those for the thicker ( $B = 1.5$  inch) specimens. This result is consistent with expectations based on both R-curve philosophy and plane-stress fracture behavior generally. However, the results for the 4T specimens in Table IX do not support this contention. That is, these 4T specimen results show that there is no effect of specimen thickness on  $K_{IC}$  behavior. Accordingly, these results on the effects of specimen thickness relative to  $K_{IC}$  behavior are inconclusive.

The lack of a consistent trend relative to the influence of thickness on  $K_{IC}$  behavior is apparently related to the local variations in fracture toughness for the 50-ksi A572 Grade 50 steel. That is, the local variations in fracture toughness are apparently of greater consequence in relation to  $K_{IC}$  than are the resulting differences in behavior between the  $B = 1.5$ -inch and  $B = 0.5$ -inch-thickness specimens. Thus, the local variation in fracture toughness for the 50-ksi A572 Grade 50 steel presently tested is apparently large enough to mask the true effect of specimen thickness on  $K_{IC}$  behavior.

2E. Local Variation in Fracture Toughness. The discussion in section 2A describes the normal, expected plane-stress fracture behavior ( $K_{IC}$  increasing with increasing  $a_0$ ). The discussions in sections 2B and 2C indicate that real scatter in  $K_{IC}$  values (or

true deviations from the expected systematic behavior) can be caused by the repeatability of a given test procedure or by differences between the load-control and displacement-control test procedures, or by both. Further deviations from the normal dependence of  $K_C$  on crack length,  $a_o$ , can also be caused by local variation in the fracture toughness of the material.

The summary of the present  $K_C$  results given in Figures 8 and 15 shows that when each of the two test procedures used is considered separately, the normal  $K_C$  behavior was usually observed, except for the results discussed below. That is, at each of the -40 F, +40 F, and +72 F test temperatures, the  $K_C$  values obtained for the 4T specimens were, as expected, higher than the corresponding  $K_C$  values for the 2T specimens for those tests conducted with the load-control procedure. Likewise, the  $K_C$  results for the 7C specimens were higher than the corresponding values for the 4C specimens for the tests conducted with the displacement-control procedure.

Exceptions to this normal  $K_C$  behavior occurred with each of the test procedures at -40 F, Table V. Specifically, for the B = 1.5-inch plate tests conducted with the displacement-control procedure, Figure 8 shows that a higher  $K_C$  value occurred for the 4C specimen ( $K_C = 102 \text{ ksi } \sqrt{\text{inch}}$  or  $112 \text{ MNm}^{-3/2}$ ) than for the 7C specimen ( $K_C = 57 \text{ ksi } \sqrt{\text{inch}}$  or  $63 \text{ MNm}^{-3/2}$ ). Similarly, for the B = 0.5-inch plate tests conducted with the load-control procedure, Figure 15 shows that a higher  $K_C$  value occurred for the 2T specimen

( $K_C = 316 \text{ ksi } \sqrt{\text{inch}}$  or  $348 \text{ MNm}^{-3/2}$ ) than for the 4T specimen ( $K_C = 150 \text{ ksi } \sqrt{\text{inch}}$  or  $165 \text{ MNm}^{-3/2}$ ). That is, for each test procedure at the -40 F temperature, a higher  $K_C$  value would normally be expected for the larger specimen (because of the larger initial crack length,  $a_0$ ), whereas a lower  $K_C$  value was actually obtained. These anomalous results—that is, these two inversions in the expected behavior—represent real  $K_C$  scatter and may be related to local variations in the fracture toughness of the A572 Grade 50 steel tested.

Another measure of local variation in the fracture toughness of the material may be obtained from Charpy V-notch (CVN) test results. Consequently, a number of CVN specimens were obtained directly from a select number of the CT specimens used in the R-curve tests. All CVN specimens were taken as close as possible to the original fracture surface of the corresponding CT specimens and in such a manner that the notch orientation for each of the CVN specimens was identical to that for the CT specimens. Approximately 10 CVN specimens were prepared from each of 11 CT specimens and tested at +72, +40, 0, and -40 F (+22, +4.5, -18, and -40 C).

The results of the CVN tests are presented in Table X, along with the results for the corresponding CT specimens. The CVN data were obtained to establish some measure of local variation of fracture toughness rather than to establish correlation between  $K_C$  and CVN test results. Because of differences in notch acuity, strain rate, and state-of-stress, any such correlations are fortuitous.

However, the variations in local fracture toughness, as measured by testing CVN specimens, can be seen when all the CVN energy-absorption values are plotted as a function of test temperature, Figure 20. The ratios of the maximum to the minimum CVN values observed at -40, +40, and +72 F ( $K_C$  test temperatures) were 7:1, 3:1, and 2.5:1, respectively. Thus, these CVN energy-absorption values, which are not at all untypical for A572 Grade 50 steel, show a large degree of variation at the same -40 F test temperature at which the anomalous  $K_C$  results were obtained. That is, the CVN results would appear to confirm that local variations in the fracture toughness of the 50-ksi A572 Grade 50 steel tested may be responsible for the inverted  $K_C$  behaviors obtained at -40 F.

Local variations in the fracture toughness of the A572 Grade 50 steel tested can also be assessed in terms of a ductility criterion rather than an energy-absorption criterion. To illustrate, when the CVN lateral-expansion (LE) values for this A572 Grade 50 steel were plotted against the corresponding CVN energy-absorption values, the correlation was nearly 1:1, as shown in Figure 21. It can therefore be concluded that a similar large variation in fracture ductility, as measured by the LE values, occurs at the -40, +40, and +72 F test temperatures.

2F. Comparison of  $K_C$  and  $K_{IC}$  Behaviors. The variation of plane-strain fracture toughness,  $K_{IC}$ , with temperature for the present A572 Grade 50 steels has been documented in a previous

study.<sup>10)</sup> For reference, the static  $K_{IC}$  results of both the 50-ksi and 62-ksi steels are presented in Figures 22 and 23, respectively. These figures show that valid  $K_{IC}$  measurements could not be made above  $K_{IC} = 55 \text{ ksi } \sqrt{\text{inch}}$  ( $60 \text{ MNm}^{-3/2}$ ) for either of the A572 Grade 50 steels. In particular, this level of  $K_{IC}$  (measured at the  $\beta = 0.40$  line intersection) occurred at  $-160 \text{ F}$  ( $-107 \text{ C}$ ) for the 50-ksi steel and at  $-60 \text{ F}$  ( $-51 \text{ C}$ ) for the 62-ksi steel. For each steel, this  $K_{IC}$  measurement limitation was a direct result of the limited plate thickness available ( $B = 1.5$  inches). Attempts to extrapolate  $K_{IC}$  behavior to higher  $K_I$  values and temperatures well beyond  $\beta = 0.40$  may lead to erroneous conclusions, particularly when, as for the present steels, only a small portion of the  $K_{IC}$  transition has been established. The inability to measure static  $K_{IC}$  values at the higher temperatures ( $-40$  to  $+72 \text{ F}$ ) was, of course, the primary reason for the present R-curve studies in this higher temperature range.

The  $K_C$  results for the  $B = 0.5$ -inch CT specimens of the 50-ksi steel obtained at temperatures between  $-40$  and  $+72 \text{ F}$  are consistent with the corresponding valid  $K_{IC}$  test results, Figure 22. That is, these results represent typical fracture behavior with  $K_C > K_{IC}$  at a given temperature as would be expected because of the different states of stress. Specifically, the  $K_C$  values of  $150 \text{ ksi } \sqrt{\text{inch}}$  ( $165 \text{ MNm}^{-3/2}$ ) and higher determined experimentally for the  $B = 0.5$ -inch CT specimens at  $-40 \text{ F}$  and above are in excess

of the corresponding  $K_{IC}$  values that would be estimated at the same temperatures by direct extrapolation of the valid  $K_{IC}$  results at cryogenic temperatures.

Several observations can be made when the present  $K_C$  results for the  $B = 1.5$ -inch CT specimens of both the 50-ksi and 62-ksi A572 Grade 50 steels are compared with the earlier valid  $K_{IC}$  results on the same scale, Figure 24. First, the two  $K_C$  results for the 62-ksi steel ( $K_C = 121 \text{ ksi } \sqrt{\text{inch}}$  at  $-40 \text{ F}$  and  $K_C = 365 \text{ ksi } \sqrt{\text{inch}}$  at  $+72 \text{ F}$ ) are again completely consistent with normal expectations. That is, these  $K_C$  values are each higher than the corresponding  $K_{IC}$  estimates of behavior that would result from direct extrapolation of the valid  $K_{IC}$  results at cryogenic temperatures for this A572 Grade 50 steel (i.e., typical plane-strain/plane-stress fracture behavior at a given temperature). Second, the present  $K_C$  results for the  $B = 1.5$ -inch CT specimens of the 50-ksi steel would appear to be somewhat inconsistent with corresponding estimates of  $K_{IC}$  behavior obtained by direct extrapolation to the  $-40$  to  $+72 \text{ F}$  temperature range. That is, the experimentally observed  $K_C$  values would appear to be lower than the estimated  $K_{IC}$  values expected from the extrapolation procedure. This apparent inconsistency remains to be explained.

If the  $K_C$  results obtained under load-control conditions (2T and 4T specimens) are considered alone, an apparent discrepancy still exists relative to the  $K_{IC}$  behavior. However, the extent of

such a discrepancy is far less under such circumstances than is the case when all the  $B = 1.5$ -inch CT results, including the  $K_C$  results obtained under displacement-control conditions (4C and 7C specimens), are considered simultaneously. As described in a previous section, the reasons for the differences between the  $K_C$  values resulting from the load-control and displacement-control testing methods are unknown and further complicate attempts to show compatibility of the present  $K_C$  and earlier  $K_{IC}$  behaviors.

The apparent inconsistency between the  $K_{IC}$  and  $K_C$  behaviors for the  $B = 1.5$ -inch specimens of the 50-ksi steel can be described in terms of the corresponding transition temperatures. That is, Figure 22 shows that the  $K_{IC}$  transition temperature is  $-160$  F ( $-107$  C), with the  $K_{IC}$  value increasing abruptly from 30 to 60 ksi  $\sqrt{\text{inch}}$  in this temperature region. Similarly, Figure 24 shows that the  $K_C$  transition temperature for  $B = 1.5$ -inch plate is approximately 0 F, with the minimum  $K_C$  value increasing abruptly from 100 ksi  $\sqrt{\text{inch}}$  at 0 F to 300 ksi  $\sqrt{\text{inch}}$  at +72 F. These results show conclusively that both the  $K_{IC}$  and  $K_C$  transitions (1) are quite steep, (2) occur at different temperatures, and (3) represent two entirely different levels of crack tolerance; the  $K_C$  transition, unlike the  $K_{IC}$  transition, represents unstable crack extension preceded by significant stable crack propagation ( $\Delta a$ ) which was as high as 4 inches.

The apparent inconsistency between the current  $K_C$  transition for the  $B = 1.5$ -inch plate of the 50-ksi steel and the corresponding  $K_{IC}$  transition may be explained if it can be established

that a  $K_{IC}$  shelf behavior exists, as shown schematically in Figure 25. The existence of a  $K_{IC}$  shelf behavior has been established earlier in 100-ksi-strength steel that is susceptible to temper embrittlement.<sup>19)</sup> Support for the possible existence of such a behavior in the 50-ksi A572 Grade 50 steel investigated may be obtained from three different sources: (1) J-integral concepts,<sup>21,22)</sup> (2) principles of the  $K_I$ -suppression effect,<sup>13)</sup> and (3) CVN specimen results.

The nonlinear concepts of fracture behavior offered by the J-integral and  $K_I$ -suppression concepts are necessary since the alleged  $K_{IC}$  shelf behavior appears to occur well above that level which can be measured validly under LEFM conditions with the B = 1.5-inch plate available ( $K_{IC} = 55 \text{ ksi } \sqrt{\text{inch}}$  at  $\beta = 0.40$  intersection, Figure 22). These nonlinear concepts have been used to re-analyze the earlier invalid  $K_{IC}$  results which exhibit increasingly more severe  $K_I$ -suppression effects for increasing temperatures above -120 F (-85 C), Figure 22. In summary form, reanalysis of the 3-point bend tests conducted at -120 and +72 F using J-integral concepts, on a conservative basis, indicated  $J_{IC}$  values that correspond to  $K_{IC}$  values of 130 and 200  $\text{ksi } \sqrt{\text{inch}}$  (143 and 220  $\text{MNm}^{-3/2}$ ), respectively. When cognizance is taken of the fact that the  $(\frac{a}{W})$  value in these tests was 0.50 instead of the near optimum 0.80 normally suggested for J-integral tests (a condition that would lead to values that are optimistic by about 20%), the adjusted  $K_{IC}$  values at -120 and +72 F are approximately 100 and 160  $\text{ksi } \sqrt{\text{inch}}$  (110 and 176  $\text{MNm}^{-3/2}$ ),

respectively. These conservative J-integral calculations indicate a gradual increase in  $K_{IC}$  with temperatures above -120 F, essentially a  $K_{IC}$  shelf behavior, rather than continuation of the steep  $K_{IC}$  transition established for lower  $K_I$  values at -160 F.

Reanalysis of the same invalid  $K_{IC}$  results using  $K_I$ -suppression effect concepts, described in detail elsewhere,<sup>13)</sup> lends additional support to the possible existence of a  $K_{IC}$  shelf behavior. In particular, it has been shown earlier<sup>13)</sup> on a 70-ksi yield-strength steel that the apparent  $K_Q$  value is suppressed to a value of 1/2 the true  $K_{IC}$  value when  $\left(\frac{K_Q}{K_{I,Gub}}\right) = 1.00$ , a condition which occurs when the test-specimen dimensions (W, B, and a) are only 1/10 of those required for a valid  $K_{IC}$  result under LEFM conditions. Table XI shows that this condition of  $\left(\frac{K_Q}{K_{I,Gub}}\right) = 1.00$  would occur at approximately -80 F (-62 C) and that the corresponding "corrected" or true  $K_{IC}$  value would be about 148 ksi  $\sqrt{\text{inch}}$  (163 MNm<sup>-3/2</sup>)--the correction being achieved by multiplying the observed  $K_Q$  value of 74 ksi  $\sqrt{\text{inch}}$  by a factor of 2.0. Furthermore, because the  $\left(\frac{K_Q}{K_{I,Gub}}\right)$  ratio for all the invalid  $K_{IC}$  tests is approximately the same (increasing only gradually from 0.94 at -120 F to approximately 1.10 or so at +72 F), these  $K_I$ -suppression effect results complement the J-integral results in providing strong indications that a  $K_{IC}$  shelf behavior occurs for the 50-ksi yield-strength A572 Grade 50 steel over the temperature range -120 to +72 F. These estimated  $K_{IC}$  behaviors obtained from both the  $K_I$ -suppression effect and J-integral concepts are summarized in Table XII.

Additional evidence in support of the  $K_{IC}$  shelf behavior but of a less direct nature can be seen from the results of CVN specimen tests. Specifically, Figure 26 shows that whereas the concept of a double shelf or double transition in the CVN energy-absorption behavior is only marginally observable under dynamic loading conditions ( $\dot{\epsilon} \approx 10^{+1} \text{ sec}^{-1}$ ), such behavior is clear and unmistakable under the same static loading conditions ( $\dot{\epsilon} \approx 10^{-5}$  to  $10^{-4} \text{ sec}^{-1}$ ) used for the present  $K_C$  tests. Although these results are presented in terms of an energy criterion, the same double-shelf or double-transition behavior can also be seen in terms of a ductility criterion, Figure 27, and, are further confirmed by considerations of fracture appearance (percent shear) behavior as well. These latter considerations also verify the existence of the upper shelf at temperatures of +30 F and higher (100% shear behavior) for the statically-tested specimens in Figures 26 and 27.

The present investigation represents the only currently known attempt to simultaneously evaluate both the  $K_{IC}$  and  $K_C$  behaviors of an intermediate-strength structural steel in a comprehensive manner. The preceding results from analysis by J-integral and  $K_I$ -suppression effect concepts as well as the quantitative CVN test results are all consistent in indicating the possible existence of a  $K_{IC}$  shelf behavior (slight positive slope) for the 50-ksi yield-strength A572 Grade 50 steel. Such behavior would clearly resolve the apparent anomaly between the present  $K_C$  transition for the  $B = 1.5$ -inch plate and the lower temperature  $K_{IC}$  transition behavior found earlier. While such a  $K_{IC}$  shelf

behavior has not been established beyond question, it is a consistent result from state-of-the-art application of nonlinear analysis techniques that are still undergoing intensive development. Because specimen thicknesses from  $B = 10$  to 25 inches would be required to establish the  $K_{IC}$  shelf behavior under valid LEFM conditions (valid  $K_{IC}$ ), it is clear that positive verification must await specifically designed tests conducted with subsize specimens and similar, but more refined, nonlinear analysis techniques in the future.

2G. Reservations Concerning Present R-Curve Results.

Descriptions of present specimen behavior above have indicated that the initial, stable crack extension for high levels of fracture resistance was such that a shear lip started to form directly from the tip of the original fatigue crack at the specimen surface, as illustrated in Figures 11, 17, and 18. That is, in the early stages of stable crack extension, the crack at the specimen surface is inclined at some angle,  $\theta$ , relative to the anticipated, flat crack-extension plane. In a later stage of stable crack extension, a full-slant fracture will develop through the specimen thickness ( $B$ ) with a resulting crack plane that is oriented at an angle of 45 degrees (0.785 rad) to the anticipated, flat crack-extension plane.

While these two different types of deviation from a flat crack-extension plane are typical for any material under true plane-stress conditions, their existence introduces additional complexities into the analysis. Specifically, a flat crack plane that is perpendicular to the applied stress ( $\sigma$ ) corresponds to the most

common type of mode I deformation of the crack, characterized by the  $K_I$  parameter. Deviations from such a flat crack plane introduce additional mode II and III deformation of the crack, described by the corresponding stress-intensity components  $K_{II}$  and  $K_{III}$ .

In all R-curve and  $K_C$  studies, including the present investigation, mode I deformation of the crack is dominant and is the only one considered in the analysis (K calculations). Such consideration of mode I alone persists even when additional mode II and III deformation components may also occur as a result of deviations from a flat crack plane. The influence of such additional modes of loading on the  $K_C$  value (calculated on the basis of mode I alone) is currently unknown. This unknown influence forms the basis of the reservations extended in relation to the accuracy of the present  $K_C$  values, particularly those for  $K_C \geq 150$  to 200 ksi  $\sqrt{\text{inch}}$  (165 to 220  $\text{MNm}^{-3/2}$ ) which required analysis by the COS method. However, as can be seen from the next section of this paper, the concern over the precision of such high  $K_C$  levels for a  $\sigma_{ys} = 50$ -ksi steel is more of an academic rather than practical nature. That is, once a behavior corresponding to a  $\left(\frac{K_C}{\sigma_{ys}}\right)$  between 2.0 and 3.0 is achieved, extensive crack tolerance ( $a_{cr}$ ) is automatically guaranteed under applied elastic stress levels ( $\sigma_D \leq \sigma_{ys}$ ).

### 3. Significance of Present R-Curve Results

3A. Critical-Flaw-Size Calculations. The significance of  $K_{Ic}$  values and  $K_C$  values derived for a given crack length (a) from a single R-curve has been discussed in concept earlier.<sup>20)</sup> For

such determinations, the parameter of ultimate interest is the critical flaw size ( $a_{cr}$ ) required to cause fracture instability under the same material and test conditions ( $T$ ,  $\dot{\epsilon}$ , and  $B$ ) used to measure the specific  $K_{Ic}$  or  $K_c$  value. The specific  $a_{cr}$  value is further related to the level of the design stress,  $\sigma_D$ , relative to  $\sigma_{ys}$  for a given specimen or structural geometry. A normalized plot showing the general relationship of  $a_{cr}$  to such parameters for a large center-cracked tension (CCT) specimen subjected to uniform tension is presented in Figure 28. Because of the normalized basis of the plot, Figure 28 can be used to calculate  $a_{cr}$  values for a CCT specimen of any material ( $\sigma_{ys}$ ) for which valid fracture-mechanics results ( $K_{Ic}$ ,  $K_{Id}$ ,  $K_c$ ,  $K_{IscC}$ ) are available under the loading rate, temperature, and state-of-stress of interest.

The specific  $K_c$  results of the current study have been summarized earlier in Figures 8 and 15. The minimum values corresponding to the bottom of the  $K_c$  scatter band for each set of results in Figures 8 and 15 can be translated into corresponding minimum values of  $a_{cr}$  for a CCT specimen with the aid of Figure 28. On the basis of the test results obtained in this study, it can be shown, Table XIII, that the minimum values of  $a_{cr}$  for 1.5-inch-thick CCT specimens subjected to a design stress,  $\sigma_D$ , equal to 3/4 the yield strength,  $\sigma_{ys}$ , are 0.58, 5.22, and 22.9 inches (14.7, 133, and 580 mm) at -40, +40, and +72 F, respectively. Table XIII also shows that at the same -40, +40, and +72 F temperatures and for the same ratio of  $\sigma_D/\sigma_{ys}$ , the minimum values of  $a_{cr}$  for B = 0.5-inch

CCT specimens are 4.06, 16.2, and >32.7 inches (103, 410, and >830 mm), respectively.

Because of the nature of the calculation for a CCT specimen, the  $a_{cr}$  value represents only half of the total central crack length. That is, the total critical crack length for a CCT specimen is  $2a_{cr}$ . When this is taken into account, the above results show with one exception that the total critical crack length ( $2a_{cr}$ ) corresponding to the minimum fracture behavior for each of the different combinations of plate thickness (B) and test temperature is at least 7 times the plate thickness ( $2a_{cr} \geq 7B$ ).

The single exception is for the B = 1.5-inch plate at -40 F, for which the total critical crack length is on the same order as the plate thickness ( $2a_{cr} = 1.16$  inches or  $29.5$  mm  $\cong B$ ). However, this calculation is based on a single data point ( $K_c = 57$  ksi  $\sqrt{\text{inch}}$  or  $63$  MNm<sup>-3/2</sup> for a 7C specimen) of doubtful representation, as discussed earlier. That is, if a more representative minimum behavior for this condition is on the order of  $K_c = 100$  ksi  $\sqrt{\text{inch}}$  ( $110$  MNm<sup>-3/2</sup>), as was indicated by a duplicate specimen test ( $K_c = 103$  ksi  $\sqrt{\text{inch}}$  or  $113$  MNm<sup>-3/2</sup> for a 7C specimen at -40 F, Table V), the corresponding  $a_{cr}$  value in Table XIII would be 1.80 inches (46 mm). In such a case the associated total critical crack length would be on the order of two and a half times plate thickness ( $2a_{cr} = 3.6$  inches or  $91$  mm  $\cong 2.5B$ ). However, an insufficient number of tests were conducted in the present study to assess the most typical behavior on a statistical basis.

The above  $a_{cr}$  values are based on using a criterion of  $\sigma_D = 3/4 \sigma_{ys}$  for the minimum  $K_c$  behavior, Table XIII. If calculations of  $a_{cr}$  are desired on the basis of a different  $\sigma_D$  criterion for the same minimum  $K_c$  behavior, they may be obtained quickly with the use of Figure 28. This same figure can also be used to obtain similar  $a_{cr}$  results for the median or maximum  $K_c$  behavior by using the corresponding center and top portions of the scatter bands shown in Figures 8 and 15.

3B. Application of Results to Structures. The  $a_{cr}$  values cited in the previous section are applicable to structures in direct proportion to the extent that the assumptions used in the basic calculation are satisfied. That is, the cited  $a_{cr}$  values are directly applicable for a structural configuration in which plane-stress conditions exist and the conditions of a large CCT specimen subjected to a remotely applied uniform stress ( $\sigma_D$ ) equal to  $3/4$  the yield strength are approximated. While differences in the nature of the stress (bending as opposed to tension) can be handled analytically,<sup>23,24</sup> the requirements of plane-stress conditions and large planar dimensions for the structural component are mandatory.

An example of the applicability of the  $a_{cr}$  values can be given in terms of a typical structural member, such as a large H-beam (girder) with typical thicknesses for both the flange and the web on the order of  $1/2$  to  $1-1/2$  inches. Specifically, the

$a_{cr}$  values cited would have application for through-thickness cracks located in the web of such a beam, where plane-stress conditions would exist. However, the same  $a_{cr}$  values would have no application for partial-thickness cracks (PTC) emanating from the top surface of the tension flange of the beam (such as would occur at the base of a cover plate due to fatigue), since the stress state at this location is primarily one of plane strain.

These same stress-state (plane stress) and structural (large planar dimensions) requirements are necessary for the interpretation of essentially all R-curve measurements, since such measurements intrinsically deal with materials exhibiting high levels of crack tolerance. In turn, high levels of crack tolerance under plane-stress conditions imply the existence of either very large critical flaws ( $a_{cr}$ ) under low levels of elastic stress ( $\sigma_D \leq 1/2 \sigma_{ys}$ ), Figure 28, or high  $K_c$  levels that translate, for short cracks ( $a$ ), into large values of the corresponding critical crack-tip plastic zone,  $r_p$ , under the action of high elastic stress ( $1/2 \sigma_{ys} \leq \sigma_D \leq \sigma_{ys}$ ). In either case, containment of such values within a large elastic-stress field is necessary before  $a_{cr}$  calculations can be valid (a fundamental principle of LEFM). Accordingly, to accomplish this containment for plane-stress conditions, large planar dimensions relative to the thickness,  $B$ , are necessary for either a specimen or a structural element.

The useful life of an H-beam subjected to load fluctuations is essentially completed when a PTC crack on the tension surface

penetrates partially through the flange. As described above, the stress state for cracks located in this tension-flange region is essentially one of plane strain—whether it is analyzable in terms of current LEFM concepts ( $K_{Ic}$ ) or not. Consequently, measurements of plane-stress fracture resistance, such as obtained with either R-curve or direct  $K_c$  measurements, have no meaning in relation to the useful life or the load-carrying capacity of such a beam.

Such plane-stress measurements would only have application in predicting the  $a_{cr}$  value at which catastrophic fracture of the H-beam would occur. For all the conditions investigated in the present study, complete failure of this type would occur only after the crack (1) had penetrated completely through the tension flange by fatigue, and (2) had subsequently propagated into the web to a crack length many times the web thickness ( $a_{cr} \gg B$ ). However, since the useful structural life of such an H-beam is expended after the first stage of fatigue-crack propagation (a condition requiring perhaps 40 to 50 years in most structural applications such as bridges), and either the H-beam is replaced or the entire structure retired from service at this point, it is academic to conjecture about the possible nature of a catastrophic fracture event that will not occur. However, the level of confidence that such an event will not occur in service can be measured in terms of the extent to which  $a_{cr} > B$ , when the appropriate material and test conditions ( $T$ ,  $\epsilon$ , and  $B$ ) have been taken into account. It is in

this indirect sense of assessing structural integrity that measurements of plane-stress fracture resistance (R-curve and  $K_{IC}$  measurements) can be beneficial when applied to structural components in service.

### Summary and Conclusions

Specific R-curve results were obtained on two different heats of ASTM A572 Grade 50 steel over the temperature range -40 to +72 F (-40 to +22 C) by using a total of 24 CT specimens. Of this total, 14 specimens had in-plane dimensions corresponding to 2T and 4T specimens and were tested under load-control conditions; the remaining 10 specimens had in-plane dimensions corresponding to 4C and 7C specimens and were tested under displacement-control conditions. Twenty-two (22) of the specimens tested were of a 50-ksi yield-strength A572 Grade 50 steel, and the two (2) remaining specimens were of a 62-ksi yield-strength A572 Grade 50 steel. Both 1.5-inch-thick and 0.5-inch-thick (38 and 12.7 mm) specimens were evaluated from the 50-ksi steel; the two specimens of the 62-ksi steel were both 1.5 inches thick. All specimens were tested under static loading conditions ( $\dot{\epsilon} \approx 10^{-5} \text{ sec}^{-1}$ ). The current study represents the first known attempt to evaluate the R-curve behavior of a high-strength structural steel. The specific results obtained from this pioneer study can be summarized as follows:

1. A steep transition was observed in the plane-stress fracture behavior for the  $B = 1.5$ -inch specimens of the 50-ksi steel, with minimum  $K_{IC}$  values of 57, 155, and 318 ksi  $\sqrt{\text{inch}}$  (63,